

# CONTROLLING CHAOS AND NONLINEAR DYNAMIC ANALYSIS OF A TWO-AXIS RATE GYRO WITH FEEDBACK CONTROL MOUNTED ON A SPACE VEHICLE

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 $\check{S}_{\lambda}(t)$ 

**Routh-Hurwits** 

Hopf

#### **Abstract**

An analysis is presented of a two-axis rate gyro subjected to linear feedback control mounted on a space vehicle that is spinning with uncertain angular velocity  $\tilde{\mathcal{S}}_{\lambda}(t)$  about its spin of the gyro. For the autonomous case in which  $\tilde{\mathcal{S}}_Z$  is steady, the stability analysis of the system is studied by Routh-Hurwits theory. For the non-autonomous case in which  $\tilde{S}_z$  is sinusoidal function, this system is a strongly nonlinear damped system subjected to parametric excitation. By varying the amplitude of sinusoidal motion, periodic and chaotic responses of this

parametrically excited nonlinear system are investigated using the numerical simulation. The results, Symmetry-breaking bifurcations, period-doubling bifurcations, and chaotic behavior of the system are observed by various numerical techniques such as phase portraits, Poincaré maps, average power spectra, and Lyapunov exponents. In addition, chaotic motions of this system can be suppressed and changed into regular motions by a suitable constant motor torque.

Keywords: Rate Gyro, Bifurcation, Chaos

#### 二、**Introduction**

A number of studies over the past few decades have shown that chaotic phenomena are observed in many physical systems that possess both non-linearity and external excitation [1]. The nonlinearity of a system, through the various system parameters, exhibits a variety of nonlinear behaviors including jump phenomenon, multiple attractors, subharmonic vibrations, symmetry breaking-bifurcations, period-doubling bifurcations, crisis and chaos [2]. In addition, a symmetry-breaking bifurcation occurring before a period-doubling bifurcation, and the appearance of chaos amidst a cascade of period-doubling bifurcations have been observed in driven damped pendulums or Duffing's oscillators by MacDonald and Räty [3]. In a gyroscopic system, a single-axis rate gyro mounted on a space vehicle free to move in various ways also exhibits complex nonlinear and chaotic motions. The nonlinear nature and chaotic motion of a single-axis

rate gyro were investigated by Ge[4] when the vehicle is spinning sinusoidally with respect to the spin axis of the gyro. This system is characterized by parametric excitation and exhibits complex nonlinear phenomena in the presence of sinusoidal excitation, including subharmonic vibrations, Hopf bifurcation, symmetry-breaking bifurcations, a series of period-doubling bifurcations, and chaos. In practice, chaotic motions are undesirable. Ge[5] used resonant parametric perturbations to change a chaotic motion into a regular one.

In this paper, an analysis is presented of a two-axis rate gyro subjected to linear feedback control mounted on a space vehicle that is spinning with uncertain angular velocity  $\tilde{\mathcal{S}}_{\lambda}(t)$  about the spin of the gyro. Here, Routh-Hurwits theory [6] is applied to analyze the stability of the autonomous case in which  $\tilde{S}_z$  is steady. For the non-autonomous case in which  $\tilde{S}_z$  is sinusoidal function, a number of numerical techniques are used to detect the existence of symmetry-breaking bifurcations, perioddoubling bifurcations, and chaos of the parametrically excited nonlinear system. The natures of the periodic and chaotic motions are shown in phase plane diagrams, Poincaré maps and average power spectra. The qualitative bifurcation diagrams, parametric diagrams and quantitative Lyapunov exponents in parametric space are also computed to determine the values of bifurcation points as well as chaos onset. In addition, chaotic motions of this system can be suppressed and changed into regular motions by a suitable constant motor torque.

### 三、**Numerical Simulations and Discussion**

 We consider the model of a two-axis rate gyro mounted on a space vehicle as shown in Fig. 1. Let *X*, *Y, Z* be a set of axes attached to the platform and  $\lt$ ,  $\lt$ ,  $\lt'$  be gimbal axes. The differential equations of a two-axis gyro with feedback control are

 $\int_{a}^{\pi}$  + 2 $r_1$ , *a*<sup>1</sup> +  $k$ , +  $r_2$ ,  $\hat{\psi}$  +  $NF_1$ ,  $\hat{\psi}$ ,  $\hat{\psi}$  = 0,  $\vec{F}$  + 2*S*<sub>1</sub>  $\vec{F}$  + *S*<sub>2</sub> $\vec{W}$  - *S*<sub>3</sub> $\vec{F}$  + *NF*<sub>2</sub> $(\mu, \mu, \phi) = 0$ where  $k=1, \, \sqrt{a} = d_{\ell}/d\ell$ ,  $\sqrt{d} = d\ell/d\ell$ ,  $\sqrt{d}f$ ,  $\sqrt{d}F_1(\ell, \ell, \ell)$ and  $NF_2(\mu, \mu, \phi)$ , shown in Appendix A.

With the system parameter *f* varied, the system results obtained by numerical integration in the phase planes, Poincaré maps, average power spectra, bifurcations and Lyapunov exponents. Hopf bifurcation occurs when the parameter *f*≈15.4, the original equilibrium point becomes unstable and a period-2*T* stable symmetric limit cycle arises as shown in Fig.2, where  $T=2\pi/5$ . A system with a symmetric nonlinear function can undergo either a symmetry-breaking bifurcation for the symmetric solution of the system or a period-doubling bifurcation for the asymmetric solution of the system. When *f*≈29.5, a symmetry-breaking bifurcation occurs. After this bifurcation, the original stable period-2*T* attractor becomes unstable, a pair of stable period-2*T* attractors arise and invert each other as shown in Fig.3 where  $f$ = 31.5. As the parameter *f* increases further across *f*≈32, a stable periodic orbit appears with double the period of the original orbit, thereby indicating a period-doubling (flip) bifurcation. When the parameter is increased, a cascade of flip bifurcations occurs and leads to the onset of chaos. At *f*≈34, the chaotic attractor abruptly disappears and a period-6T symmetric orbit appears, as shown in the phase plane and average power spectrum (Fig.2,4).

To investigate bifurcation further, a Poincaré plane was used to display the bifurcation diagram, which shows Poincaré fixed points  $x_p$  plotted against the system parameter *f*. The Hopf bifurcation, symmetry -breaking bifurcation, and period-doubling bifurcation are clearly shown. As the system parameter *f* is gradually increased through the parametric space, the bifurcation diagram obtained shows different types of bifurcations and chaos (Fig.5). The Hopf bifurcation at *f*≈15.4, the symmetry-breaking bifurcation at *f*≈29.5, and the period-doubling bifurcation at *f*=32, as observed earlier. To investigate the periodic and chaotic motions in the bifurcation diagram further, the phase planes, Poincaré maps, and power spectra are used. After a cascade of period-doubling bifurcations, the dual response becomes chaotic rather than periodic for *f*=32.5. When *f*=33, conjunction of the two inverse chaotic attractors creates a larger attractor. With the

parameter increased, a large-amplitude chaotic motion appears in the phase plane, Poincaré map, and power spectrum as shown in Fig.6, where  $\neq 36.3$ . The power spectrum of a chaotic motion is a continuous board spectrum.

To confirm the chaotic dynamics, a quantitative Lyapunov-exponent spectrum was performed. The algorithm for calculating the Lyapunov exponents was developed by Wolf et al. [7]. A spectrum of the largest Lyapunov exponent as a function of the parameter *f* is shown in Fig.7. As one of the Lyapunov exponents is positive, the motion is characterized as chaotic. When at least one Lyapunov exponent  $\lambda_i = 0$  exists, motions are not stationary. For periodic motions, the Lyapunov exponents are non-positive and include only one zero Lyapunov exponent, while one negative exponent becomes zero when one type of periodic motion bifurcates to another.

Physically, chaos may be desirable or undesirable, depending on the application. In this case, we used a feedback constant control torque with the assistance of the Lyapunov exponent calculations to bring the system from a chaotic regime to a regular. For changing the parameter  $k$  form 0.5 to 1.5, there are the bifurcation diagram and the spectrum of the largest Lyapunov exponents *lmax* as the function of the stiffness coefficient *k* in Fig.8. As  $\lambda_{max} < 0$  for the suitable  $k$ , the system is periodic.

### 四、**Conclusions**

In this paper, a two-axis rate gyro with sinusoidal velocity about its spin axis *<sup>Z</sup>* exhibits the nonlinear characteristic of both sin, cos function and parametric excitation when the parameter is varied. For the autonomous case in which  $\check{S}_z$  is steady, the stability conditions were derived by the Routh-Hurwitz criterion. A variety of parametric studies were performed to analyze the behavior of periodic attractors route to chaos via distinct bifurcations by using the numerical simulations. The behaviors of a symmetry-breaking precursor to perioddoubling bifurcations and a cascade of period-doubling route to chaos occurred in

this system. The occurrence of the chaotic motion of the full system is also detected by calculating bifurcation diagrams, power spectral diagrams and Lyapunov exponents. In addition, we consider a suitable feedback constant force torque to suppress chaos in the system by computing Lyapunov exponents.

#### 五、**References**

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## **Appendix A**

 $\Delta F_1(_{x},\mathbf{W},\mathbf{1}) = (A+B_1-C_1)(-\stackrel{\circ}{W}{}_{C_2}\stackrel{\sim}{S}_{Z} \stackrel{\sim}{S}_{n} + (-\stackrel{\sim}{S}_{n}^2 \stackrel{\sim}{W}{}_{S_1+S_1C_2}{}^2 \stackrel{\sim}{S}_{L}^2 + 2\stackrel{\sim}{W}{}_{C_2}\stackrel{\sim}{S}_{L}$  $(\tilde{S}_{n}c_1)c_1)/[(A+A_1)\tilde{S}_n^2]$ -Hc(- $\tilde{S}_n\vec{W}$ c<sub>1</sub>- $s_1c_2\tilde{S}_{Z}$ +  $\vec{W}\tilde{S}_n)/[(A+A_1)\tilde{S}_n^2]$ -( $\vec{W}$ c<sub>2</sub>  $\times \check{S_{\mathrm{Zt}}}\check{S_{\mathrm{n}}}$  +  $s_{2}\check{S_{\mathrm{n}}}\check{\check{S_{\mathrm{Zt}}}})$  /  $\check{S_{\mathrm{n}}}^{2}$  $\mathcal{N}\!\mathit{F}_2 ($ <sub>s</sub>,w,t)=((s<sub>1</sub><sup>2</sup>  $\int_a^1 c_2 \check{S}_{Z} \check{S}_n A_2$ -k<sub>2</sub>Ws<sub>1</sub><sup>2</sup>-  $\hat{W}$   $d_2 \check{S}_n s_1^2) C_1 + (-\check{S}_{Z}^2 s_2 A_2 + \check{S}_{Z}^2)$  $\int_{\mathscr{A}} \tilde{S}_{n}A_{2}c_{2}(A+A_{1})+(2C_{1}s_{1}\stackrel{\rightharpoonup}{W}\int_{\mathscr{A}} \tilde{S}_{n}^{2}A_{2}-C_{1}s_{1}c_{2}\stackrel{\rightharpoonup}{S}_{Z_{1}} \tilde{S}_{n}A_{2}+(-\tilde{S}_{Z_{1}}\stackrel{\rightharpoonup}{W}\tilde{S}_{n})$  $\times A_2 + \tilde{S_{Z}}^2 s_2 A_2) c_2 C_1 c_1) c_1 + (\int\limits_{J'} \tilde{S_{n}} C_1 s_1^2 + \int\limits_{J'} \tilde{S_{n}} A_2 + (s_2 \tilde{S_{Z}} A_2 - \int\limits_{J'} \tilde{S_{n}} A_2) c_1)$  $\times$ Hc+(kzW+(-s'i ]  $\vec{S}_{Z} \vec{S}_{n} A_{2}+s_{1}^{2} s_{2} \vec{S}_{z} + \vec{W}_{d_{2}} \vec{S}_{n} + C_{1} s_{1}^{2}$  ]  $c_{2} \times \vec{S}_{Z}$  $\int_{\mathcal{A}} + (-\tilde{S}_{Z}^{2} s_{2} + \tilde{S}_{Z} \int_{\mathcal{A}} \tilde{S}_{n}) c_{2}(A+A_{1}) + (-2s_{1} \stackrel{\dagger}{W} \int_{\mathcal{A}} \tilde{S}_{n}^{2} A_{2} + 2C_{1}s_{1} \stackrel{\dagger}{W} \int_{\mathcal{A}} \tilde{S}_{n}^{2} +$  $_{(s_1c_2}\check{S}_nA_2-C_1\times s_1c_2\check{S}_n)}\check{S}_{Zt}+(-k_2w+c_2\check{S}_{Zt''}^{\ \ T}\check{S}_nA_2-\hat{W}d_2\check{S}_n+(\check{S}_{Zt}^{\ \ T})$  $\propto$ s2- $\tilde{\mathcal{S}}_{Zt}$ nn $\tilde{\mathcal{S}}_{Zt}$ nn $\tilde{\mathcal{S}}_{Zt}$  (2) C1) C1+(s2 $\tilde{\mathcal{S}}_{Zt}$ nnn $\tilde{\mathcal{S}}_{Zt}$ nnnn $\tilde{\mathcal{S}}_{Zt}$  $-S_1^2 \int\limits_{\mathscr{B}}^{\mathscr{B}} \tilde{S}_{\mathscr{A}} \, \tilde{S}_{\mathscr{n}} \rangle_{C_2} + (-2 S_1 \iint\limits_{\mathscr{B}}^{\mathscr{B}} \int\limits_{\mathscr{B}} \tilde{S}_{\mathscr{n}}^2 + S_1 C_2 \, \tilde{\mathbf{S}}_{Z} \, \tilde{S}_{\mathscr{n}} + C_2 \tilde{S}_{Z \, \mathscr{B}} \, \int\limits_{\mathscr{B}} \tilde{S}_{\mathscr{n}} C_1 \rangle_{C_1} ) (A_1 + A_2)$  $+ B_1$ ))( $A + B_1$ ))/[ $((A + B_1)c_1^2 + C_1s_1^2 + A_2)(A + B_1 + A_2)\tilde{S}_n^2$ ] where  $s_1 = \sin s$ ,  $s_2 = \sin t$ ,  $c_1 = \cos s$ ,  $c_2 = \cos t$ , etc.  $\tilde{S}_{\text{Zt}} = \sin \tilde{S}$ ,  $\tilde{S}$  $= \check{\mathcal{S}_{\!f}}/\check{\mathcal{S}_{\!n}},$ *w*& *Zt*  $= d\check{S}_{\mathrm{Zt}}/d\varphi$ 

Fig.1 A two-axis rate gyro.



Fig.2 Two inversion-symmetric attractors: a period-2*<sup>T</sup>* attractor for  $f=18$ , a period-6*T* attractor for  $f=$ 34 where the symbols  $+$  and  $\cdot \times$  indicate one period-*T* of  $\check{S}_Z = f \sin \check{S} \check{\mathcal{I}}$ .



Fig.3 A dual period-2T attractor for  $\neq 31.5$ . <sup>4</sup> Average Power Spectral Density



Fig.4 An average power spectrum of Fig. 4 for  $f=34$ .



Fig.5 The bifurcation diagram.



Fig.6(a) A symmetric chaotic attractor for *f*=36.3. Poincare Maps



Fig.6(b) A symmetric chaotic attractor for *f*=36.3.



Fig.6(c) An Average power spectrum for *f*=36.3.



Fig.7 The largest Lyapunov exponents.



Fig.8 The bifurcation diagram and the largest Lyapunov exponent as a function of *k*.